



# Experiment to Test Low Concentration Volcanic-Ash Ingestion by a Jet Engine

October 22, 2015  
Anchorage, Alaska

**John Lekki, Jack Hoying, Mike Venti, Donald Simon, Marianne Guffanti, Andrew W. Phelps, John Fisher, Allan van de Wall, Fred Smith**



# VIPR Partnership



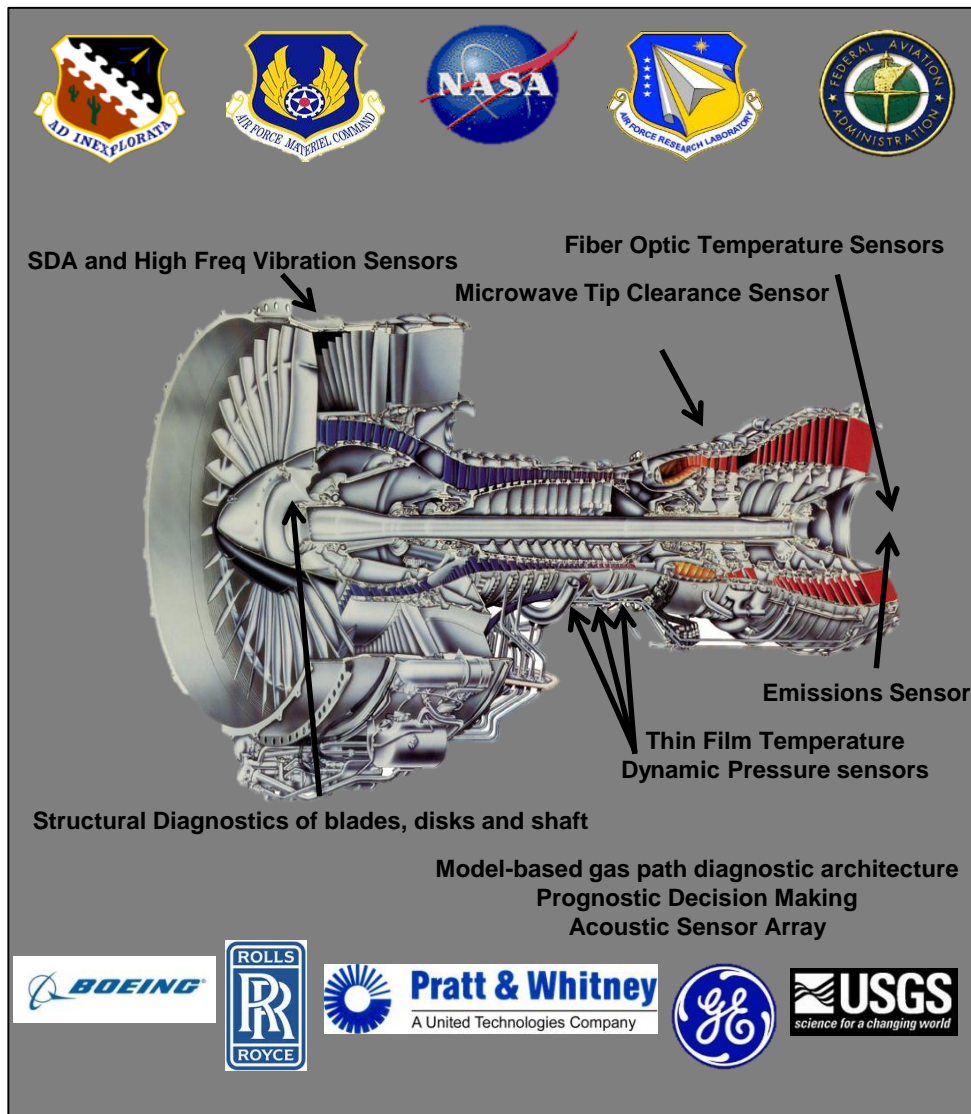
**PARTNERSHIPS MAKE IT POSSIBLE**

## VIPR Stakeholders Contribution

- Direct: funds, equipment, material
- In-kind: resources for project success

## Installed Engine Ground Tests

- Large bypass transport engine
- Baseline as-is engine operations
- Induced mechanical faults
- Induced gas path faults
- Simulate ash-air laden exposure
- Characterize degradation







# Engine Health Management



## Existing Aircraft

- Engines are highly reliable....however
- Engine malfunctions contributing to accidents and incidents do occur
- Ground-based testing may not identify problems occurring in-flight
- EHM is limited due to the harsh environment operational conditions
- Malfunction examples include
  - uncontained rotor failures
  - in-flight engine shutdowns
  - restricted thrust response
- Examples of underlying causes include
  - environmental effects such as volcanic ash and ice ingestion
  - turbomachinery damage
  - controls and accessory faults



Propulsion System Malfunction combined with Inappropriate Crew Response Accidents



# Engine Health Management Potential Technology Benefits



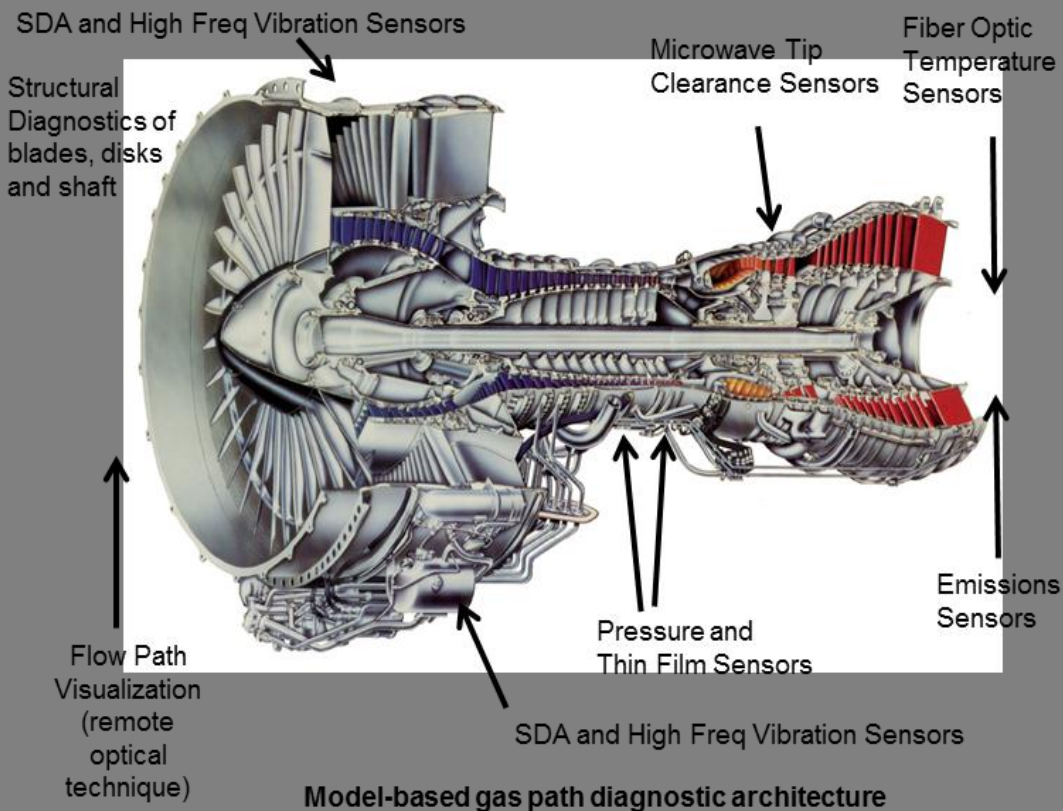
- Safety
  - Improved engine life prediction
  - Improved real-time failure diagnosis
- Performance and Reliability
  - Improved engine performance
  - Reduced maintenance costs
  - Increased asset availability
  - Improved fuel efficiency
  - Increased range



# VIPR 3



**Goal: Determine capability of advanced detection, diagnostic and prognostic systems to characterize engine performance, and identify fault modalities, during rapid engine degradation caused by the ingestion of volcanic ash.**





# Ground Testing Overview



## VIPR 1 (DECEMBER 2011): PERIPHERAL SENSORS

- SUCCESSFULLY INTEGRATED EXPERIMENTAL TECHNOLOGIES
- SELF DIAGNOSTIC ACCELEROMETER
- MODEL BASED DIAGNOSTICS
- EMISSIONS SENSORS

## VIPR 2 (JULY 2013): INTEGRATED CORE SENSORS

- SUCCESSFULLY INTEGRATED EXPERIMENTAL TECHNOLOGIES
  - MICROWAVE BLADE TIP CLEARANCE SENSOR
  - THIN FILM PRESSURE SENSORS
- DETECTED & CHARACTERIZED INDUCED FAULT IMPACTS



## VIPR 3 (2015): INTEGRATED ADVANCED & MATURED SENSORS

- INDUCED VOLCANIC ASH INGESTION - RAPID ENGINE DEGRADATION
- DETERMINED CAPABILITY OF ADVANCED DETECTION
- CHARACTERIZED ENGINE PERFORMANCE [*DIAGNOSTIC & PROGNOSTIC*]
- IDENTIFIED FAULT MODALITIES





# Test Completion



- **Advanced Instrumentation installed on engine and initial testing and check out occurred during hanger integration test February 2015**
- **Engine: Commercial variant that was significantly modified to demonstrate the VIPR technology and additional sensor suite installed on wing of an aircraft May 2015**
- **Testing**
  - Test Start Date: June 16, 2015
    - Combined Systems Test (CST)
    - Bleed Air Extraction & Sampling System (BAESS)
    - Engine Health Management (EHM)
    - Volcanic Ash Environment (VAE)\*
  - Test Completion: August 5, 2015
- **5 Volcanic Ash Ingestion Tests completed**



| Date      | Target Concentration (mg/m <sup>3</sup> ) | Daily Run Time (min) | Daily Ash Ingested (kg) |
|-----------|---|----------------------|-------------------------|
| 28-Jul-15 | 1   | 90                   | 0.730                   |
| 29-Jul-15 | 1   | 68                   | 0.549                   |
| 31-Jul-15 | 1   | 269                  | 2.156                   |
| 04-Aug-15 | 10  | 175                  | 11.017                  |
| 05-Aug-15 | 10  | 235                  | 14.465                  |

Public Release CLEARED on 22 Oct 2015

Originator Reference Number: RQ-15-789 Case Number: 88ABW-2015-5137



# Overview

## Vehicle Integrated Propulsion Research (VIPR) III

### GREAT NEWS!



E15 Volcanic ash cloud in Iceland  
April 2010



GLASSY BUILD-UP

SHEDDING

#### **FOCUS:** "ASH (*NON-VISIBLE/VISIBLE SPECTRUM*) THRESHOLD DECISION-POINT DEBATE"

- EVALUATED FLIGHT DECISION POINT CONCENTRATIONS RATES OF **LOW (1 MG/M3)** & **HIGH (10 MG/M3)**  
— US GOVERNMENT & MANUFACTURERS TEAM - SUMMER 2015, EDWARDS AFB
- **PREDICTED ENGINE DEGRADATION** WITHIN **1HR @ LOW**; & **RED-LINE BREACH** (*ENGINEER- SET -MARGIN THRESHOLD*) @ **3HR HIGH**

**RESULTS:**  **VERIFIED COMPRESSOR BLADE EROSION**

**TURBINE MOLTEN ASH GLASSY BUILD-UP**

**ASH SHEDDING**

**ENGINE PERFORMANCE DEGRADATION @10HR**

**14HR CUMULATIVE TEST NO RED-LINE BREACH**

**WHAT'S NEXT:** DATA ANALYSIS; ENGINE ANALYTICAL CONDITION INSPECTION ...*BEYOND VIPR III*

#### **BOTTOM LINE:**

- 1ST CONTROLLED VOLCANIC ASH EXPOSURE EXPERIMENT CONSISTENT WITH FLIGHT SAFETY POLICY
- MATURED KEY TURBINE ENGINE TECHNOLOGIES RELEVANT TO AEROSPACE COMMUNITY



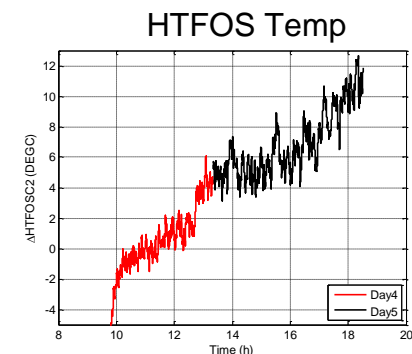
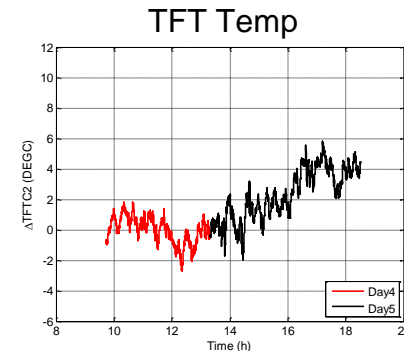
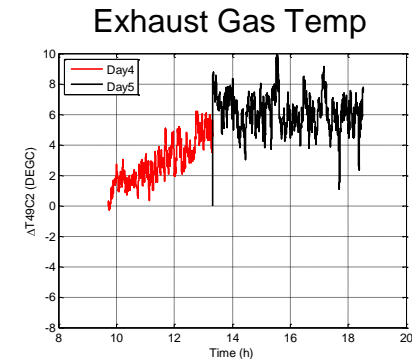
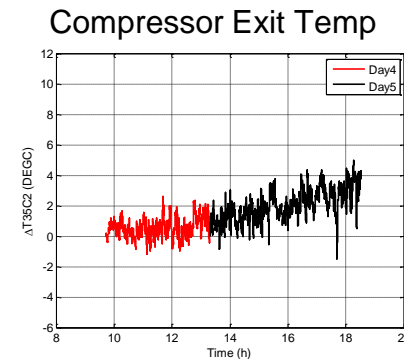
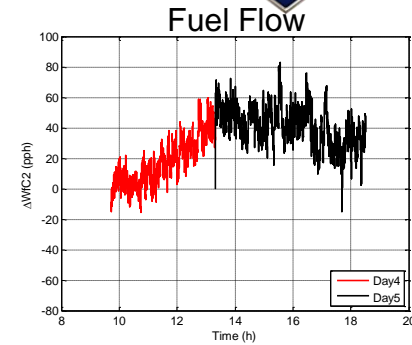
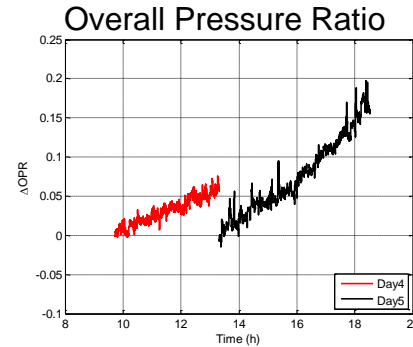


# Gas Path Measurement Data



## RESULTS AND SIGNIFICANCE (Preliminary)

- Five (5) days of volcanic ash ingestion testing
  - Days 1, 2, and 3 ran low concentration ash ingestion
  - Days 4 and 5 ran higher concentration ash ingestion
- No significant engine performance variations were observed during low concentration ash runs
- On high ash concentration run days, discernable performance trend changes were observed in overall pressure ratio (OPR), fuel flow, compressor exit temperature, and exhaust gas temperature.
- Advanced sensor data tracks performance changes observed elsewhere in engine
  - High Temperature Fiber Optic Sensor (HTFOS) trends with exhaust gas temperature
  - Thin Film Thermocouple (TFT) trends with compressor exit temperature





# High Temperature Fiber Optic Sensor



## OBJECTIVE

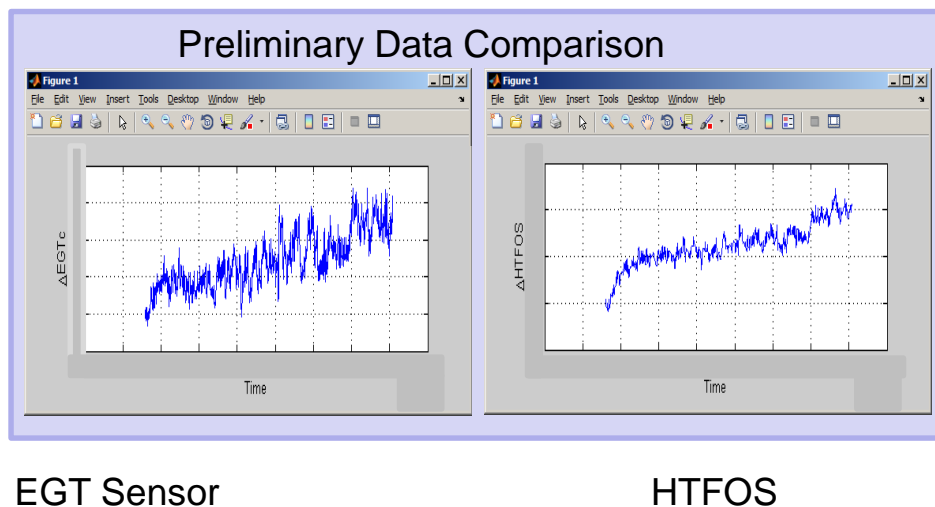
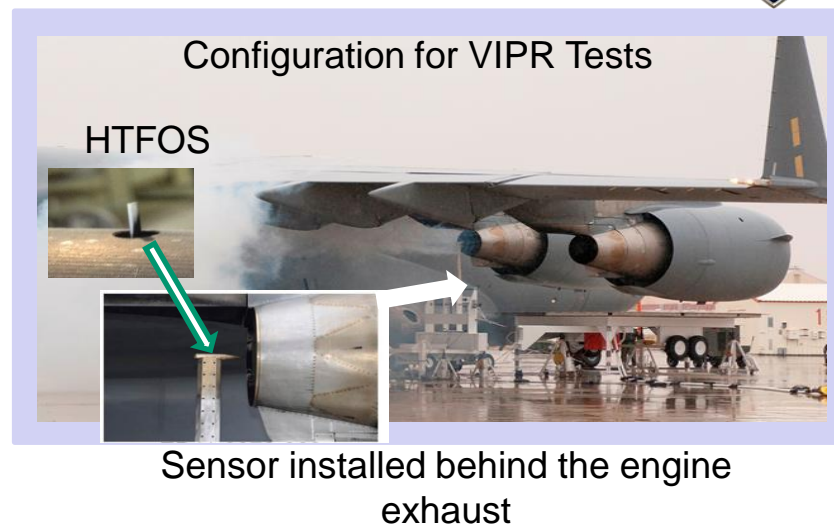
- Develop and evaluate silica-based fiber optic sensor suitable for operation at the engine exhaust temperatures

## APPROACH

- Evaluate the sensor performance in the engine exhaust plume under various engine operating and ash ingestion conditions.
- Compare results with the EGT sensor data

## RESULTS AND SIGNIFICANCE

- Shown capable of withstanding thermal and vibrational environments of the jet engine exhaust
- Significance of the sensor is immune to EMI and EMP, has high resistance to chemicals, low noise, signal fidelity, low maintenance
- Preliminary results show lower noise than existing EGT sensors





# Emissions Sensor Suite Volcanic Ash



## OBJECTIVE

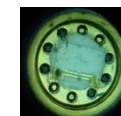
- Demonstrate the ability to diagnose engine faults and performance loss effects using emission sensor array

## APPROACH

- Install sensor suite in engine exhaust

## RESULTS AND SIGNIFICANCE (Preliminary)

- Emissions Sensor Array Monitored Engine Emissions during Days 1-3, 5 of Ash runs
- Days 1-3: Candidate “Steady-State” Emissions Parameter Identified
- Day 5 Deviation in “Steady State” Emissions Profile Suggestive of Change in Engine State During Heavy Volcanic Ash Deposition
  - Not Presently Explainable By Reference to Other Engine Parameters



CO sensor

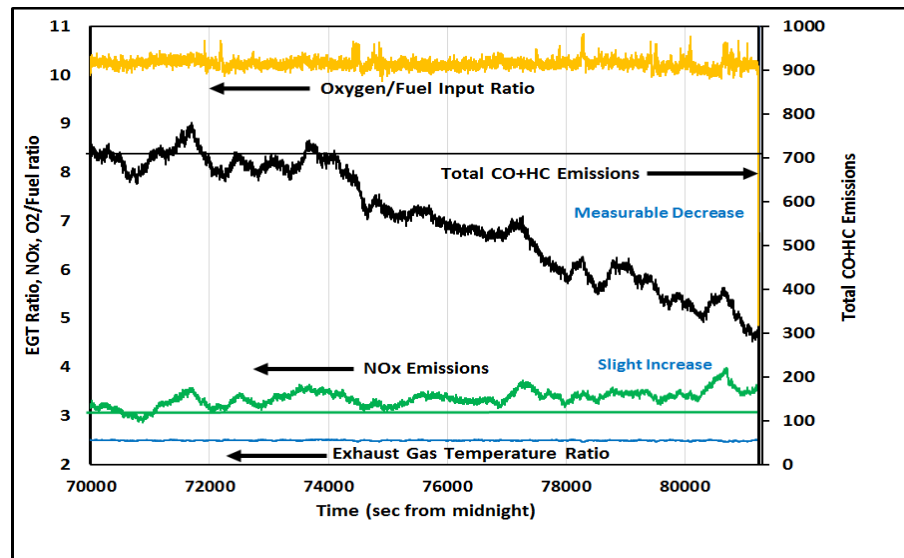


Sensor probe

Sensor probes in emissions testing rig manifold



Preliminary Emissions Data – Patterns Established/ Deviations Observed Suggesting Possible Volcanic Ash Effects







# Test Results



| Information targeted for release                            |  |
|---|--|
| <b>Hardware degradation Specifications</b>                  | % of hardware deemed repairable and unserviceable via borescope inspections as a function of time and locations.       |
|   | Description of failure mode.   |
| <b>Measured parameters</b><br>(Absolute and percent change) | EGTcorr Corrected exhaust gas temperature  |
|   | EPR Engine pressure ratio  |
|   | N1C2 Corrected low rotor speed   |
|   | N2C2 Corrected high rotor speed  |
|   | WFcorr Corrected fuel flow   |
|   | PBcorr Corrected burner pressure   |
| <b>Derived Parameters</b><br>(percent change from baseline) | Time history of engine data.   |
|   | W2corr Corrected station 2 core flow (estimated)   |
|   | OPR Overall pressure ratio   |
|   | Op. Lines LPC and HPC operating lines (estimated)  |
|   | Surge Line HPC low power surge line (estimated)  |
| <b>Other Engine Data</b>                                    | Ambient conditions during test. Oil analysis test results.   |
| <b>Ash Analysis</b>   | Composition going in, % deposited in different locations (if obtained), composition of what is deposited (if obtained) |
| <b>Air circuit information</b>                              | % of airflow circuit   |

Note all information must be reviewed to confirm there are no data release restrictions

Public Release CLEARED on 22 Oct 2015

Originator Reference Number: RQ-15-789 Case Number: 88ABW-2015-5137



# Report Plan



**Goal is to have a report released next summer**

- 1) Executive Summary
- 2) Background/Motivation
- 3) VIPR3 VAE Test Plans and Test Configuration
- 4) Description of VIPR3 VAE Test Execution
- 5) VIPR3 VAE Test results
  - a) Performance results
  - b) Shedding Results
  - c) Observations of engine degradation
- 6) Summary / Conclusions / Recommendations / Caveats



# VIPR 1, 2 and 3 Summary

## Testing complete: highly successful



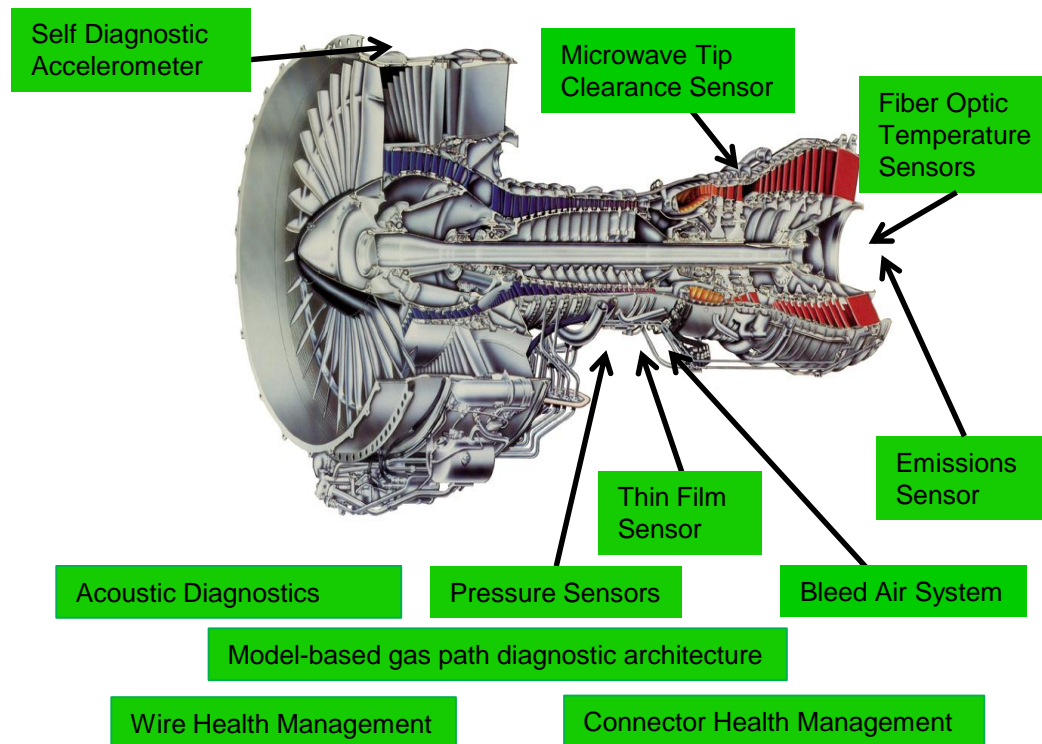
### Test Objectives:

Demonstrate capability of advanced health management technologies for detecting and diagnosing incipient engine faults before they become a safety impact and to minimize loss of capability

### Approach:

Perform on wing engine ground tests

- Normal engine operations
- Seeded mechanical faults
- Seeded gas path faults
- Accelerated engine life degradation through volcanic ash ingestion testing



VIPR 2 Test completed in July 2013

VIPR 3 Test completed in August 2015